BEHAVIOR AND DIAGNOSTIC ANALYSIS FOR AIRCRAFT ENGINES

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Marie Skłodowska-Curie Project, MOIRA – H2020 20/06/2023



Why Aircraft Engine Diagnostics?











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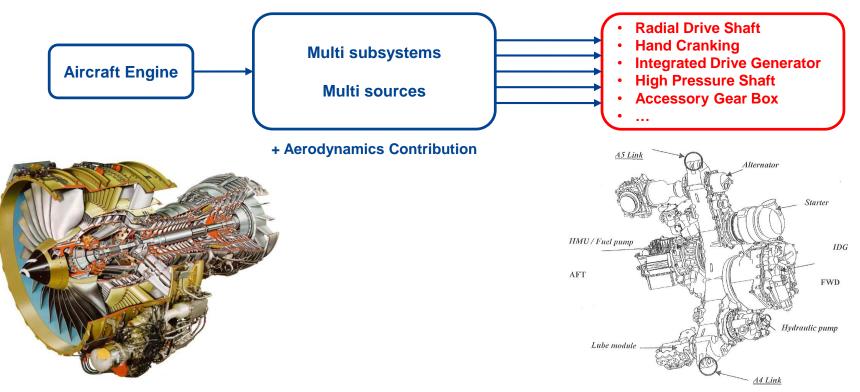


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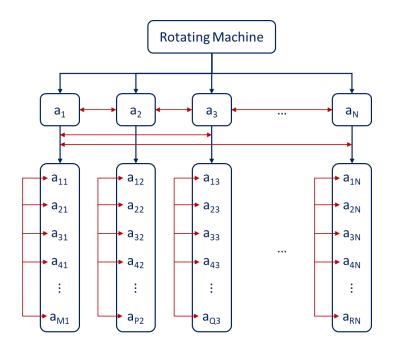
An Aircraft Environment





Mechanical Fault Characteristics

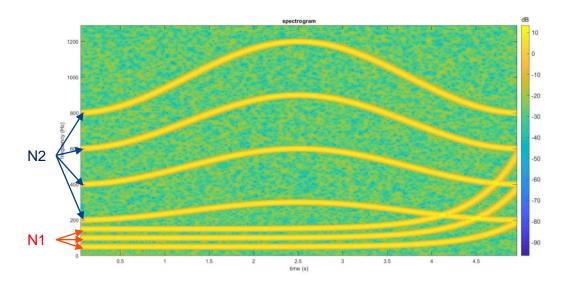
- Each subsystem consists of several mechanical component
- Each mechanical component rotates at given frequency
- Each corresponding fault frequency is related to the component base frequency
- Several subsystems + Several harmonics = Several interferences
- What if the component base frequency is not constant?





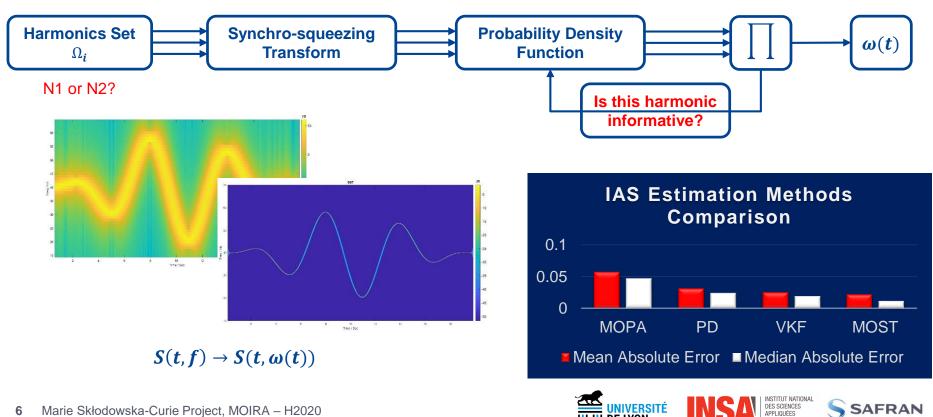
Instantaneous Angular Speed (IAS) Estimation

- Frequency characteristics varying over time
- N1 or N2?
- Characterization of $\omega(t)$
- Repetitive profile due to harmonics
- Noisy Harmonics
- Signal to noise ratio and interferences





Multi Order Synchro-squeezing Transform

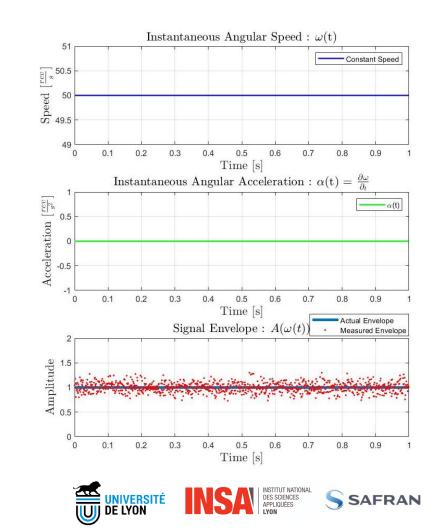


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Components Signatures

- Model: $y(t) = A(t)e^{j\theta(t)} + v(t)$
- For Stationary Signals :
- $\frac{\partial A(\omega(t))}{\partial \omega} = \frac{\partial A(t)}{\partial t} = \varepsilon(t)$

Resulting an uncertainty error variance of : σ_{ε}^2



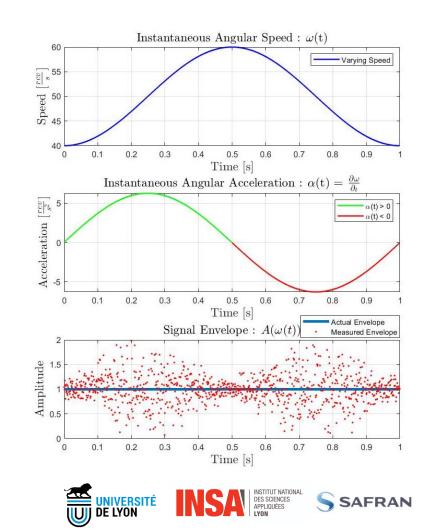
Components Signatures

- Model: $y(t) = A(t)e^{j\theta(t)} + v(t)$
- For Non Stationary Signals :

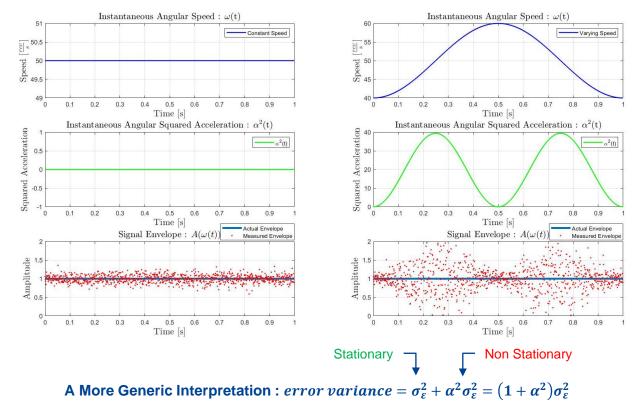
•
$$\frac{\partial A(\omega(t))}{\partial \omega} \neq \frac{\partial A(t)}{\partial t} = \varepsilon(t)$$

 $\rightarrow \frac{\partial A(\omega)}{\partial t} = \alpha \times \varepsilon(t)$

Resulting an uncertainty error variance of : $\alpha^2 \sigma_{\epsilon}^2$



Components Signatures





Discussion and Conclusion

- Upcoming works
- Deterministic component extraction
- Envelope analysis
- Component Diagnosis
- Subsystem health state evaluation
- Fault localization and identification
- Industrial Applicability
- Kinematics and Frequency band a priori knowledge
- Estimation of both noise and uncertainty errors











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